

Gravitational 2-body problem : II

- bound vs. unbound orbits
 - elliptical shape of bound orbits
 - period of elliptical orbits:
Kepler's laws
-

Without loss of generality (choice of coordinates) we set $E_0 = 0$ in our orbit formula :

$$u(\theta) = \frac{1}{r(\theta)} = \frac{1}{r_0} (1 + \epsilon \cos \theta)$$

The eccentricity ϵ is a dimensionless ~~non-negative~~ parameter, where $\epsilon = 0$ corresponds to a circular orbit. Since the sign of ϵ can be

flipped by the rotation $\theta \rightarrow \theta + \pi$,
we may assume $E \geq 0$.

It is interesting to see how
the total energy

$$E = \frac{L_z^2}{2\mu} \left[\left(\frac{du}{d\theta} \right)^2 + u^2 \right] - Au$$

$\frac{Ar_0}{2}$

depends on the parameters r_0 and
 E of our orbit:

$$E = \frac{L_z^2}{2\mu} \left[\frac{E^2}{r_0^2} \sin^2\theta + \frac{1}{r_0^2} + \frac{2E}{r_0^2} \cos\theta + \frac{E^2}{r_0^2} \cos^2\theta \right] - \frac{A}{r_0} (1 + E \cos\theta)$$

$$= \frac{Ar_0}{2} \left[\frac{E^2}{r_0^2} + \frac{1}{r_0^2} \right] - \frac{A}{r_0}$$

$$= \frac{A}{r_0} \left[\frac{E^2}{2} - \frac{1}{2} \right] = \frac{A}{2r_0} (E^2 - 1)$$

(2)

We see that if $0 \leq e \leq 1$ the energy E is negative, and inconsistent with the scenario where the separation between the masses grows without bound (since then the energy would be purely kinetic and positive). The case $0 \leq e < 1$ therefore corresponds to bound orbits.

A simple calculation shows the shape of a bound orbit is an ellipse. From the orbit equation

$$\frac{1}{r(\theta)} = \frac{1}{r_0} (1 + e \cos \theta)$$

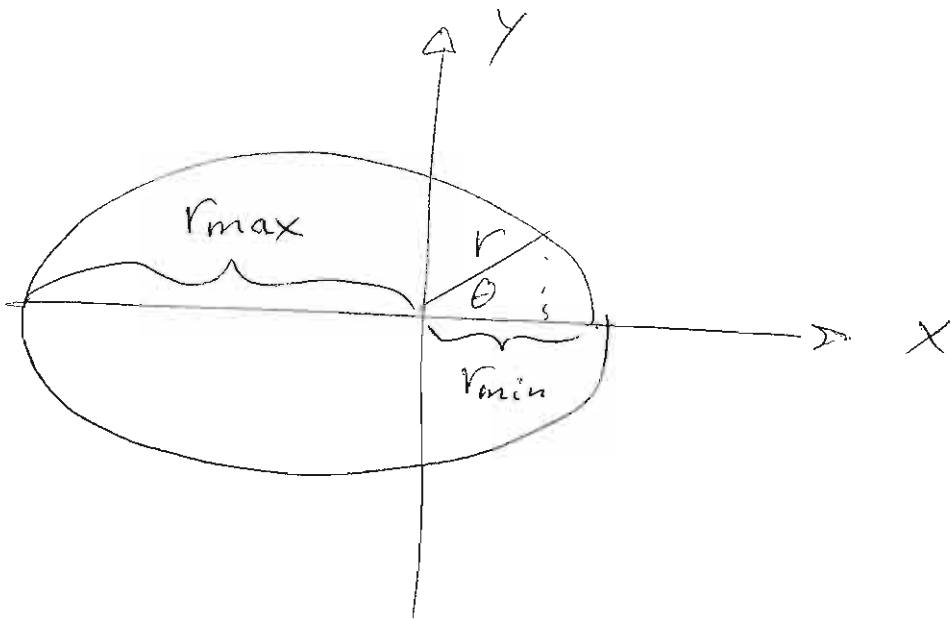
We see that when $e < 1$ the

(3)

separation $r(\theta)$ varies between the extremes

$$r_{\max} = \frac{r_0}{1-\epsilon} \quad , \quad r_{\min} = \frac{r_0}{1+\epsilon} .$$

$(\theta = \pi)$ $(\theta = 0)$



From

$$r_0 = r + r\epsilon \cos \theta$$
$$= r + \epsilon x$$

we find

$$(r_0 - \epsilon x)^2 = r^2 = x^2 + y^2$$

$$r_0^2 = (1 - \epsilon^2)x^2 + 2\epsilon r_0 x + y^2$$

(4)

$$(1-e^2)(x-x_0)^2 + y^2 = r_0^2 + \underbrace{(1-e^2)x_0^2}$$

$$x_0 = -\frac{e}{1-e^2}r_0$$

$$\left(1 + \frac{e^2}{1-e^2}\right)r_0^2 = \frac{r_0^2}{1-e^2}$$

define: $a = \frac{r_0}{1-e^2}$, $b = \frac{r_0}{\sqrt{1-e^2}}$

and x, y satisfy:

$$\frac{(x-x_0)^2}{a^2} + \frac{y^2}{b^2} = 1,$$

the equation of an ellipse with semi-major axis a , semi-minor axis b . As $e \rightarrow 1$ the ellipse becomes very elongated:

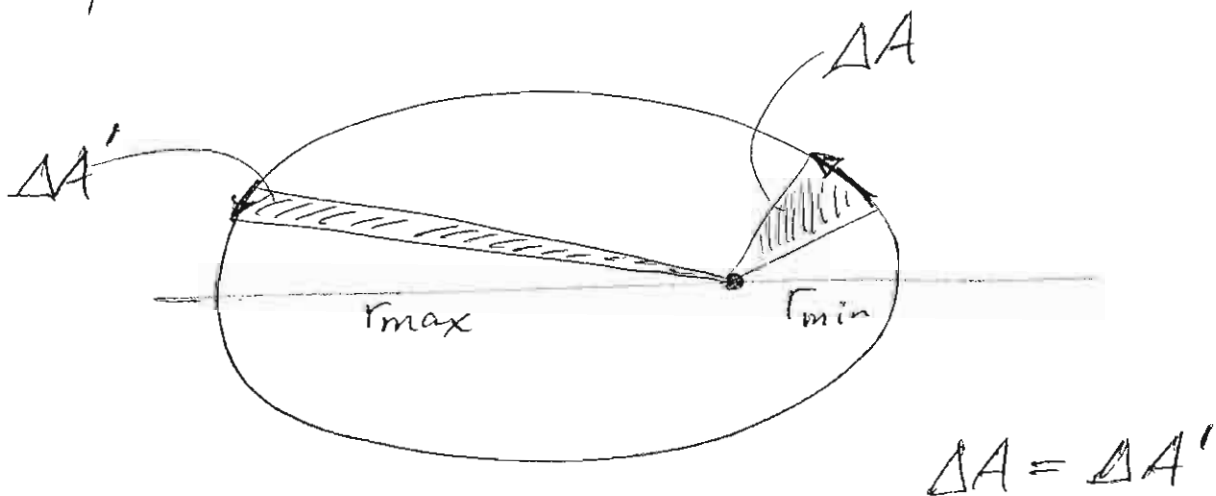
$$\frac{a}{b} = \frac{1}{\sqrt{1-e^2}} \rightarrow \infty$$

In that same limit

$$r_{\min} \rightarrow \frac{r_0}{2}, \quad r_{\max} \rightarrow \infty$$

and the orbit becomes unbound at $e = 1$. We consider the case $e > 1$ later.

Kepler's 1st law states that the "position vector sweeps out equal areas in equal time". This actually refers to the relative position vector:



This law is actually a direct consequence of the constancy of L_z :

$$\Delta A = \left(\frac{\Delta \theta}{2\pi} \right) \pi r^2 = \frac{1}{2} r^2 \dot{\theta} \Delta t$$

$$\frac{\Delta A}{\Delta t} = \frac{1}{2} r^2 \dot{\theta} = \frac{L_z}{2\mu} = \text{constant}$$

Kepler's 2nd law :

$$(\text{orbit period})^2 \propto (\text{semi-major axis})^3$$

This follows from the integral of the area derivative over one period :

$$A = \oint_0^T \dot{A} dt = \frac{L_z}{2\mu} \cdot T$$

Geometrically,

$$A = \pi a b$$

But we will express this in terms of just a :

$$b^2 = \frac{r_0^2}{1-e^2} = a^2(1-e^2)$$

$$A^2 = \pi^2 a^2 \cdot a^2(1-e^2)$$

$$= \pi^2 a^3 r_0$$

Comparing the two expressions for A^2 :

$$\frac{L_z^2}{(2\mu)^2} T^2 = \pi^2 a^3 r_0$$

$$= \pi^2 a^3 \frac{L_z^2}{A\mu} \quad (\text{previous lecture})$$

$$\Rightarrow T^2 = 4\pi^2 \left(\frac{\mu}{A}\right) a^3$$

$$= \left(4\pi^2 / G(M_1 + M_2)\right) a^3$$

(8)

It is useful to know that this law applies to arbitrary elliptic orbits, not just nearly circular ones.